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REPORT NO. 89-R-08 AFPEA PROJECT NO. 88-P-102

Robbin L. Miller

Mechanical Engineer

AUTOVON 787-3362

Commercial (513) 257-3362



QUALIFICATION TESTING OF THE COMBAT TALON II MULTISENSOR SIGNAL PROCESSOR CONTAINER

HQ AFLC/DSTZ AIR FORCE PACKAGING EVALUATION ACTIVITY Wright-Patterson AFB OH 45433-5999

August 1989

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#### MOTICE

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#### ABSTRACT

Aeronautical Systems Division, ASD/VXAL, requested assistance from the Air Force Packaging Evaluation Activity (AFPEA) to choose an off the shelf container and qualify it for the multisensor signal processor (S/P) used on Combat Talon II aircraft.

The S/P prototype container was tested at the AFPEA, HQ AFLC/DSTZ, Wright-Patterson AFB, OH 45433-5999. The container is environmentally sealed and outfitted with a humidity indicator and pressure relief valve. The container is designed to protect one S/P during worldwide shipment, storage, and handling.

The container test plan was developed to test the fragility and environmental sealing qualification requirements. The tests were conducted in accordance with Federal Test Method Standard No. 101, and Military Standard 648.

Results of the tests conducted on the prototype container show that the container provides adequate mechanical protection but only marginal environmental protection. Based on the projected operational environment, the system program office has elected to use the container.

PREPARED BY: PUBLICATION DATE:

Robbin Miller

Robbin Miller

Mechanical Engineer

AF Packaging Evaluation Activity

REVIEWED BY:

Ted Hinds

Ch, Design Branch

AFPEA

0. AUG 1989

APPROVED BY:

Charlie P. Edmonson Chief, AF Packaging

Evaluation Activity

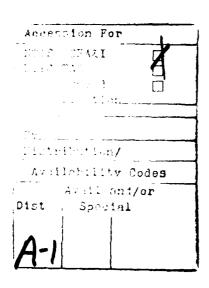
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#### INTRODUCTION

BACKGROUND: Aeronautical Systems Division (ASD/VXAL), Wright-Patterson AFB OH 45433-5000 requested assistance from the Air Force Packaging Evaluation Activity (AFPEA) to choose an off the shelf container for the multisensor signal processor (S/P) and perform qualification testing. The container chosen was a plastic multipurpose container designed by Hardigg Industries, South Deerfield, MA 01373.

<u>PURPOSE</u>: The purpose of this project was to determine if the container design will protect the contents, one S/P for Combat Talon II aircraft, during worldwide shipment, storage, and handling.

#### DESCRIPTION OF TEST CONTAINER

Two identical containers were involved in the testing. The 11214-8678-200 prototype container, now referred to as -200A or -200B were subjected to extensive testing. The sides, latches and hinges of the container were numbered counterclockwise from the forward end as shown in figure 1.

<u>Design</u>: The -200 prototype is a controlled-breathing container with a pressure relief valve and humidity indicator (see figure 2). The container is designed to limit the transmission of shocks to the S/P to 40 Gs. The container cover is permanently hinged on one side (see figure 3) and eight wing latches on the remaining sides allow quick access to the container contents without the use of tools.

<u>Construction</u>: The container is rotationally molded from a formulation of polyethylene. Two pound density polyethylene foam encapsulates the item (see figure 4). A silicone gasket provides a seal between the container base and the container cover.

## TEST OUTLINE AND TEST EQUIPMENT

Test Plan: Tests were conducted in accordance with AFPEA Test Plan 88-P-102 (see table 1). The tests used were selected to meet the qualification requirements for fragility and environmental sealing. Test methods, procedures and pass/fail criteria used were as outlined in Federal Test Method Standard No. 101 (FED-STD-101) and Military Standard 648. Any modifications to the standard procedures are noted in the test plan or the results. Each test result will specify the container used for that test, -200A or -200B.

Test Load: All tests were conducted using the S/P test load fabricated at the AFPEA. The test load weighs 89 pounds and simulates the center of gravity and the mass moment of inertia of an actual S/P.

Test Site: All testing was conducted at the AFPEA, HQ AFLC/DSTZ, Building 70, Area C, Wright-Patterson AFB OH 45433-5999. The equipment required for each test is noted in the test plan.

#### TEST PROCEDURES AND RESULTS

#### Weight Test

Test No. 1: The containers were weighed to determine weight compliance.

Results: Total tare weight of each container was 52.5 pounds. The results of this test are acceptable.

## Leak Test

Test No. 2: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2 at 0.50 psig. The vacuum retention test was conducted in accordance with FED-STD-101, method 5009.2 at 0.50 psig. The failure criteria for the test was 0.025 psig loss during a 30 minute period, after temperature and pressure stabilization.

Results: At the end of the 30 minute test period the pressure loss for container -200A was 0.024 psig and 0.021 psig for container -200B. At the end of the 30 minute test period the vacuum loss for containers -200A and -200B was 0.024 psig. The results of this test are acceptable.

# Free Fall Drop Tests (+140°F)

Test No. 3: The high temperature free fall drop tests were conducted in accordance with FED-STD-101, Method 5007.1. The height of the drops were 18 inches.

Results: Visual inspection revealed no external damage to the -200A container. A maximum of 13 Gs was obtained during the tests.

The container was opened after the free fall drop tests. Visual inspection revealed no damage to the container or the test load. The results of these tests are acceptable. See appendix 1 for detailed acceleration results.

### Leak Test

Test No. 4: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2. The test was performed at 0.50 psig. The failure criteria for the test was a 0.0125 psig loss during a 15 minute period after temperature and pressure stabilization.

Results: At the end of the 15 minute test period the pressure loss for container -200A was 0.0125 psig. The result of this test is acceptable.

# Free Fall Drop Tests (-20°F)

Test No. 5: The low temperature free fall drop tests were conducted in accordance with FED-STD-101, Method 5007.1. The height of the drops were 18 inches.

Results: Visual inspection revealed no external damage to the -200A container. A maximum of 15 Gs was obtained during the tests.

The container was opened after the free fall drop tests. Visual inspection revealed no damage to the container or the test load. However after cold conditioning the decals fell off the container. The results of this test are acceptable.

#### Leak Test

Test No. 6: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2. The test was performed at 0.50 psig. The failure criteria for the test was a 0.0125 psig loss during a 15 minute period after temperature and pressure stabilization.

Results: At the end of the 15 minute test period the pressure loss for container -200A was 0.01 psig. The result of this test is acceptable.

#### Superimposed Load Test

Test No. 7: The ambient superimposed load test was conducted in accordance with FED-STD-101, Method 5016.1. A load of 2434 pounds was placed on top of the container using load spreaders. This simulates the loading of a stack of nine containers with a safety factor of two on the bottom container.

Results: Visual inspection revealed no permanent deformation to the -200A container. The result of this test is acceptable.

### Leak Test

Test No. 8: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2. The test was performed at 0.50 psig. The failure criteria for the test was a 0.0125 psig loss during a 15 minute period after temperature and pressure stabilization.

Results: At the end of the 15 minute test period the pressure loss for container -200A was 0.011 psig. The result of this test is acceptable.

## Vibration Fatique Test

Test No. 9: The vibration fatigue test was conducted in accordance with MIL-STD-648, paragraph 5.3.2. The container was rigidly attached to the platform. A sinusoidal vibration excitation was applied in a vertical direction and cyclically swept for 7.5 minutes at 2 minutes per octave to locate the resonant frequency. Input from 5 to 12.5 Hz was at 0.125 inch double amplitude and input from 12.5 to 50.0 Hz was at 1.0 G. A 30 minute dwell test was conducted at the resonant frequency.

Results: Visual inspection revealed no damage to the -200A container or the test load. A maximum of 2.5 Gs was obtained at the resonant frequency of 17.2 Hz. The maximum transmissibility obtained was 1.7. The results of this test are acceptable.

#### Leak Test

Test No. 10: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2. The test was performed at 0.50 psig. The failure criteria for the test was a 0.0125 psig loss during a 15 minute period after temperature and pressure stabilization.

Results: The -200A container would not pressurize and a leakage rate could not be obtained. Although the container failed the leak test after the vibration fatigue test, the failure was a result of deformation to the hinges due to overall use during testing.

Anomaly: At this time container -200A was removed from the testing procedures and replaced with container -200B to finish the test plan as requested by the program office. Production containers will be equipped with latches and not hinges (see recommendations). Based on the projected operational environment, sealing ability is not a necessity.

## Vibration Fatigue Test

Test No. 9: The vibration fatigue test was repeated for container -200B in accordance with MIL-STD-048, paragraph 5.3.2. The container was rigidly attached to the platform. A sinusoidal vibration excitation was applied in a vertical direction and cyclically swept for 7.5 minutes at 2 minutes per octave to locate the resonant frequency. Input from 5 to 12.5 Hz was at 0.125 inch double amplitude and input from 12.5 to 50.0 Hz was at 1.0 G. A 30 minute dwell test was conducted at the resonant frequency.

Results: Visual inspection revealed no damage to the -200B container or the test load. A maximum of 2.6 Gs was obtained at the resonant frequency of 15.0 Hz. The maximum transmissibility obtained was 1.2. The results of this test are acceptable.

### Leak Test

Test No. 10: The pneumatic pressure test was repeated for container -200B in accordance with FED-STD-101, Method 5009.2. The test was performed at 0.50 psig. The failure criteria for the test was a 0.0125 psig loss during a 15 minute period after temperature and pressure stabilization.

Results: At the end of the 15 minute test period the pressure loss for container -200B was 0.011 psig. The result of this test is acceptable.

## Hoisting Strength Test

Test No. 11: The single ring hoisting test was conducted in accordance with MIL-STD-648, paragraph 5.8.5. The loaded container was lifted by a lift ring and suspended for five minutes.

<u>Results</u>: Visual inspection of the -200B container revealed no damage or deformation. The result of this test is acceptable.

### Leak Test

Test No. 12: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2. The test was performed at 0.50 psig. The failure criteria for the test was a 0.0125 psig loss during a 15 minute period after temperature and pressure stabilization.

Results: At the end of the 15 minute test period the pressure loss for container -200B was 0.012 psig. The result of this test is acceptable.

## Superimposed Load Test

Test No. 13: The high temperature, high humidity superimposed load test was conducted in accordance with FED-STD-101, Method 5016.1. A load of 1217 pounds was placed on top of the container using load spreaders this simulates the loading of a stack of nine containers with a safety factor of one on the bottom container.

<u>Results</u>: Visual inspection revealed a permanent deformation of the container. This deformation did not cause any damage to the S/P or impair stackability.

#### Leak Test

Test No. 14: The pneumatic pressure test was conducted in accordance with FED-STD-101, Method 5009.2. The test was performed at 0.50 psig. The failure criteria for the test was a 0.0125 psig loss during a 15 minute period after temperature and pressure stabilization.

Results: The -200B container would not pressurize and a leakage rate could not be obtained. This occurred due to the permanent deformation of the container from the high temperature, high humidity superimposed load test.

#### CONCLUSION

The -200 prototype container provided adequate mechanical protection for the contents when tested in accordance with the container test plan. However, the container provides marginal environmental protection, especially if the container will be shipped and/or stored in a stacked configuration under tropical conditions.

#### RECOMMENDATIONS

The container should have wing latches only, no hinges. Additional cushion cut outs around the pressure relief valve and homidity indicator. Decals on the containers need better adherence for cold temperature environments. Container walls need to be made stiffer for more stability. The container should not be used for long term storage.

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PACK DESCRIPTION				
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As noted below.				
TEST REF STD/SPEC NO. AND TEST METHOD OF PROCEDURE NO'S	TEST TITLE AND PARAMETERS	CONTA		INSTRU- MENTATION
1. WEIGHT TEST	•			T
	Total container weight should not be less than 52 lbs.		tainer ding sola-	Scale
2. <u>LEAK TEST</u>				· !
FED-STD-101 Method 5009.2	Pneumatic pressure at 0.50 PSIG and vacuum retention at 0.50 PSIG. Test duration to be a minimum of 30 minutes with 0.025 PSIG loss allowed after temperature stabilization.	from pressed	dition com- air	Water manometer
3. FREE FALL DROP	TESTS (HIGH TEMPERATURE +1	140°F)		İ
FED-STD-101 Method 5007.1	Free fall drop test. Condition at +140°F for not less than 24 hours. Drop height 18 inches or maximum height where container does not tilt over on side. Peak resultant acceleration shall not exceed 40Gs.	Test peri in cha Drop on of sides	mber. half and	Tri-axial accelero- meters
COMMENTS: 4				
* Remaining	impacts to be performed in	rest No.	5.	
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Susan Mughey, Mechan	nical Engineer Ted Hinds	, Chief, I	Design E	Br., AFPEA

AFALD FORM 4

#### AIR FORCE PACKAGING EVALUATION ACTIVITY (Container Test Plan) 88-P-102 CUBE (CU. FT.) QUANTITY CONTAINER SIZE (L x W x D) (INCHES) WEIGHT (LBS) DATE GROSS: ITEM: INTERIOR: EXTERIOR: 3 Jul 89 MANUFACTURER ITEM NAME Hardigg Industries LRUS CONTAINER NAME CONTAINER COST Part numbers 11214-8678-200 PACK DESCRIPTION Composite Container CONDITIONING As noted below. REF STD/SPEC TEST CONTAINER INSTRU-AND TEST METHOD OR TEST TITLE AND PARAMETERS NO. MENTATION ORIENTATION PROCEDURE NO'S LEAK TEST 4. FED-STD-101 Pneumatic pressure with Ambient 0.50 PSIG. Test duramanometer Method 5009.2 tion not less than 15 minutes with 0.0125 PSIG loss allowed after tempcrature stabilization. ROUGH HANDLING TESTS (LOW TEMPERATURE -20 F) 5. FED-STD-101 aree fall drop test. Test performed Tri-axial Condition at -20°F for in chamber. Method 5007.1 acceleronot less than 24 hours. meters Drop on half of sides and 'Drop height 18 inches or maximum height where corners, total container does not tilt four 10 f Peak resultant drops. over. acceleration shall not exceed 40Gs. LEAK TEST 6. Water FED-STD-101 Pneumatic pressure with Ambient Method 5009.2 0.50 PSIG. Test duramanometer tion not less than 15 minutes with 0.0125 PSIG loss allowed after temperature stabilization. 7. SUPERIMPOSED LOAD (Ambient temperature) FED-STD-101 At ambient temperature, Stack two Visual high, bottom inspection Method 5016.1 stack two containers with additional load on container under test. top to simulate stacking These impacts are opposite those performed in Test No. 3. APPROVED BY: PREPARED BY: Ted Hinds, Chief, Design Br., AFPEA Susan Hughey, Mechanical Engineer AFALD FORM 4 PAGE 2 OF 6 8

AFPEA PROJECT NUMBER

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Composite Container				
As noted below.				
TEST REF STD/SPEC AND TEST METHOD OR PROCEDURE NO'S	TEST TITLE AND PARAMETERS	CONT	AINER INST	
	minutes. Then dwell 30 minutes at the resonant frequency. The test may be interrupted to prevent excessive temperature rise in materials. Transmissibility shall not exceed 5 at the resonant frequency.			
© <u>TEAK TEST</u> SD-STD 101 Method 5009.2	Pneumatic pressure with 0.50 PSIG. Test duration not less than 15 minutes with 0.0125 PSIG loss allowed after temperature scabilization.	Ambient	Water manome	ter
MIL-STING STRENGT MIL-STD-648 Para. 5.8.5	Single ring hoisting test. Hoist container at one lift point and leave hanging for five minutes. There shall be no damage or permanent deformation.	Ambient	Visual inspec	tion
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REPARED BY:	APPROVED B			
usan Hughey, Mechan:			Design Br., AF	PEA
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AIR FORCE PACKAGING EVALUATION ACTIVITY  (Container Test Plan)  AFPEA PROJECT NUMBER  88-P-102										
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AFALD FORM 4

# AFPEA PROJECT NUMBER AIR FORCE PACKAGING EVALUATION ACTIVITY (Container Test Plan) 88-P-102 CONTAINER SIZE (L x W x D) (INCHES) CUBE (CU. FT.) QUANTITY DATE WEIGHT (LBS) GROSS: ITEM: EXTERIOR: INTERIOR: 3 Jul 89 ITEM NAME MANUFACTURER Hardigg Industries LRUS CONTAINER NAME CONTAINER COST Part numbers 11214-8678-200 PACK DESCRIPTION Composite Container CONDITIONING As noted below. REF STD/SPEC TEST CONTAINER INSTRU-AND TEST METHOD OR TEST TITLE AND PARAMETERS ORIENTATION PROCEDURE NO'S MENTATION NO. 14. LEAK TEST FED-STD-101 Pneumatic pressure with Ambient 'Water Method 5009.2 0.50 PSIG. Test dura-Manometer tion not less than 15 minutes with 0.0125 PSIG loss allowed after temperature stabilization. COMMENTS: APPROVED BY: PREPARED BY: Susan Hughey, Mechanical Engineer Ted Hinds, Chief, Design Br., AFPEA AFALD FORM 4 PAGE 6 of 6 12

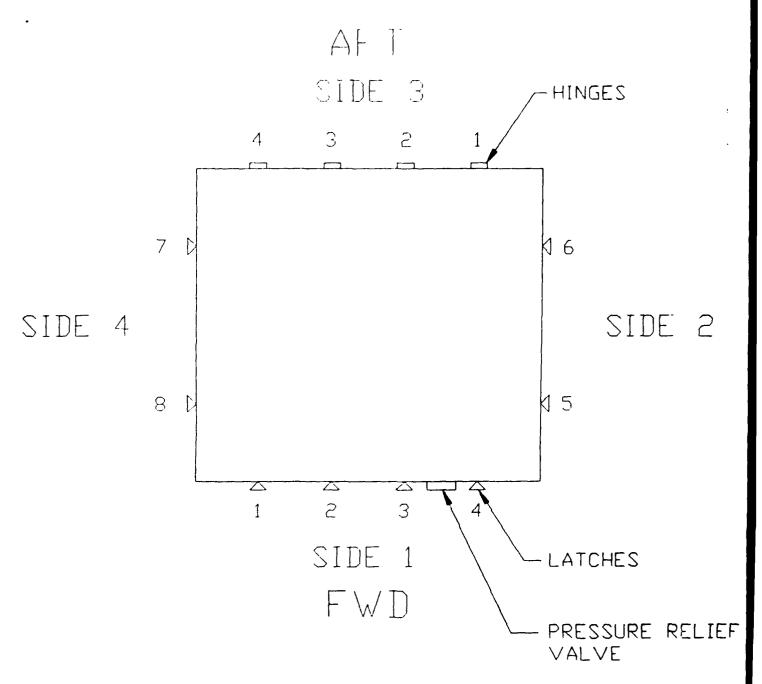


FIGURE 1. -200 Side, Latch and Hinge Numbering.

Figure 2
-200
Prototype
Container.

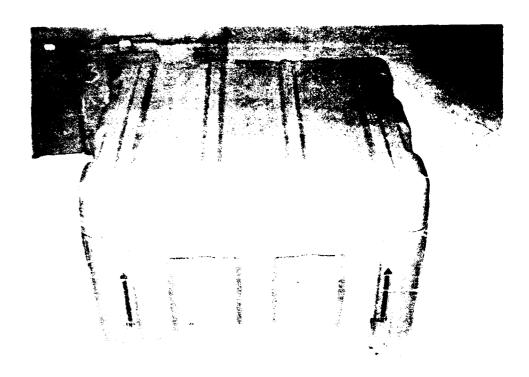
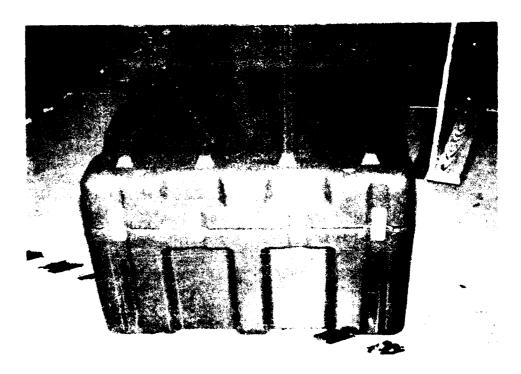


Figure 3
-200
Hinge
Placement.



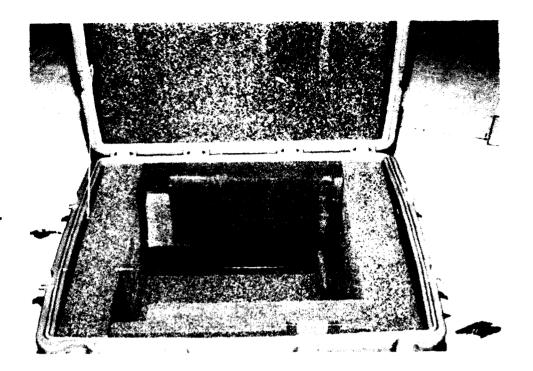


Figure 4

-200
Container
Cushioning.

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OO-ALC/DSTP Hill AFB UT 84406	1
SA-ALC/DSTP Kelly AFB TX 78241	1
SM-ALC/DSTP McClellan AFB CA 95652	1
WR-ALC/DSTP Robins AFB GA 31098	1
ASD/ALXP/SDM Wright-Patterson AFB OH 45433	1
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Commander	1
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# HIGH TEMPERATURE ROUGH HANDLING TESTS (+140°F)

Impact	Position	Accelerometer readings Resultant	(Gs)
18" free fall drop 18" free fall drop 18" free fall drop 18" free fall drop	Corner 1-2 Corner 3-4 Side 3 Side 4	<del></del>	

1. No damage to the container or the test load.

# LOW TEMPERATURE ROUGH HANDLING TESTS (-20 F)

Impact		Accelerome Position	eter readings (Gs) Resultant
18" free f 18" free f 18" free f 18" free f	fall drop fall drop	Corner 1-4 Corner 2-3 Side 1 Side 2	13 13 9 15

1. No damage to the container or the test load.

# VIBRATION FATIGUE TEST

Natural frequency 15.0 Hz

(input: 1.04 G peak, 0.125 inch double amplitude)

#### Resultant

Maximum	Acceleration (Gs	s,	peak	to	peak)	2.6
Maximum	Transmissibility	У			- '	1.2

1. No damage to the container or the test load.